

United States of America
Department of Transportation — Federal Aviation Administration
Supplemental Type Certificate

Number SA367NE

This certificate, issued to General Electric Company
Aircraft Engine Group
1000 Western Avenue
Lynn, Massachusetts
certifies that the change in the type design for the following product with the limitations and conditions therefor as specified hereon meets the airworthiness requirements of Part 25 of the Federal Aviation Regulations.

SURRENDERED

Original Product — Type Certificate Number: A52EU
Make: SAAB-SCANIA AB
Model: SF-340A

Description of Type Design Change:

Installation of General Electric CT7-5A2 engines in accordance with SAAB-SCANIA AB S-340 Maintenance Manual No. AMM SF340A 000, dated July 1, 1984, including Temporary Revision No. 71-6 dated September 12, 1984, and General Electric Installation Manual, Document No. SEI-585, Revised May 1984.

Limitations and Conditions:

1. General Electric Co. Airplane Flight Manual Supplement No. 1, dated August 1, 1985, or later FAA approved revisions is required for this installation.

2. Checking of the N₁ bottoming governor every 25 hours in accordance with SAAB-Fairchild Service Bulletin SF340-05-002, Rev. 1, dated October

18, 1984, for the General (See STC Continuation Sheet, Page 2)
This certificate and the supporting data which is the basis for approval shall remain in effect until surrendered, suspended, revoked, or a termination date is otherwise established by the Administrator of the Federal Aviation Administration.

Date of application: July 12, 1985
Date of issuance: August 1, 1985

Date issued:

Date amended:



By direction of the Administrator

Edward W. Mavila
(Signature)

for Ronald L. Vavruska, Manager
Boston Aircraft Certification Office
(Title)

Any alteration of this certificate is punishable by a fine of not exceeding \$1,000, or imprisonment not exceeding 3 years, or both.

This certificate may be transferred in accordance with FAR 21.47.

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Supplemental Type Certificate
(Continuation Sheet)

Number SA367NE

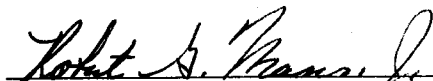
Limitations and Conditions (Continued)

2. Electric Company (GEC) Engine Model CT7-5A1, Rev. 1, dated October 18, 1984, which is also applicable to the GEC Engine Model CT7-5A2 (STC SA367NE approval) is required for this installation.
3. This installation is limited to the Dowty Rotol Model R.320/4-123-F/1 propeller.
4. The engine constant torque governing system is not approved for this installation, (Ref. Note 25 of TC Data Sheet E8NE).
5. This approval should not be extended to other specific airplanes of this model on which other previously approved modifications are incorporated unless it is determined that the interrelationship between this change and any of the other previously approved modifications will introduce no adverse effect on the airworthiness of that airplane.

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Surrender Conditions:

1. Future unsafe conditions existing in the product can result in an airworthiness directive requiring correction of the unsafe condition before further flight. If there is no entity to comply with 14 CFR § 21.99(a), required design changes, the existence of an unsafe condition might result in permanent grounding of the aircraft on which the STC is installed.
2. An FAA Export Certificate of Airworthiness will not be issued for the product after April 5, 2010.


(Signature)

Robert G. Mann
Manager
Boston Aircraft Certification Office

APR 06 2010

(Date)

-END-

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